EME antenna and Jupiter noise on 77 GHz

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EME 2014
Pleumeur-Bodou (France)

Antenna

Offset dish



Material 2.5 mm Al Diameter 2400 mm Focal length 1380 mm

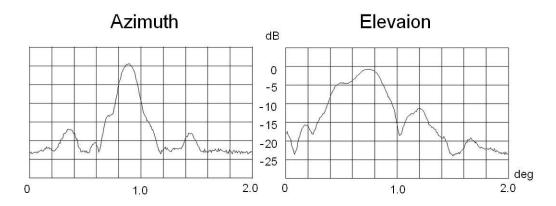
HPBW 0.11°
Gain 62.5 dBi



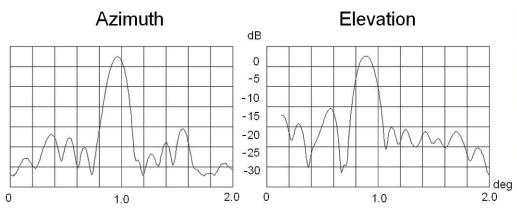
Skobelev DMH (RA3AQ)

First antenna pattern measurement

November 2010

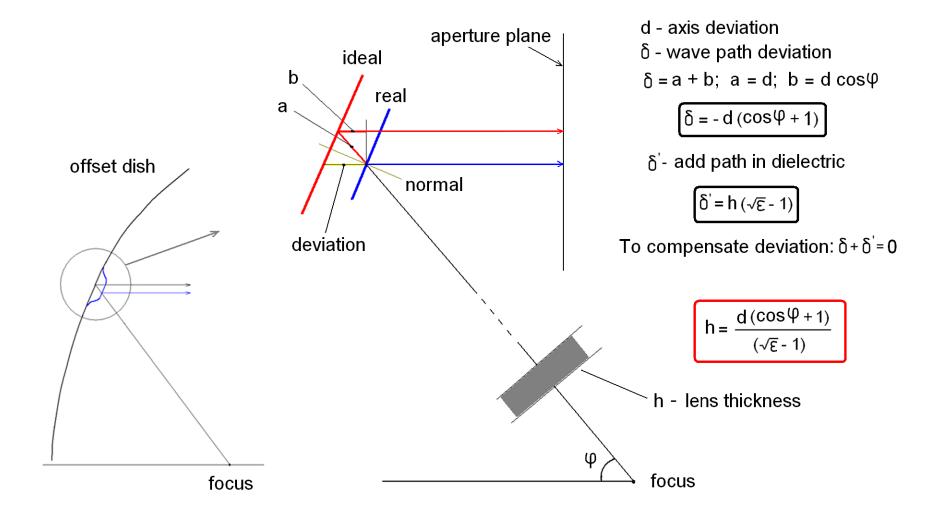


First simple dielectric lens June 2011





Dielectric lens

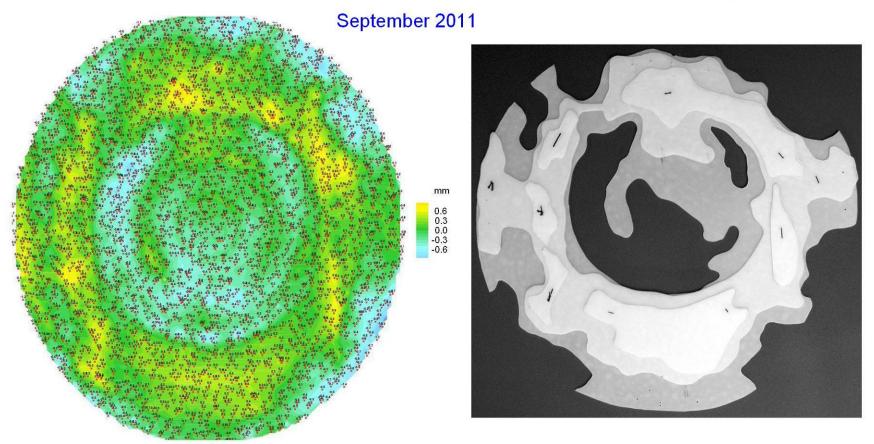


Map of deviations

RMS = 0.31 mm

Dielectric lens

Three 1 mm PTFE layers



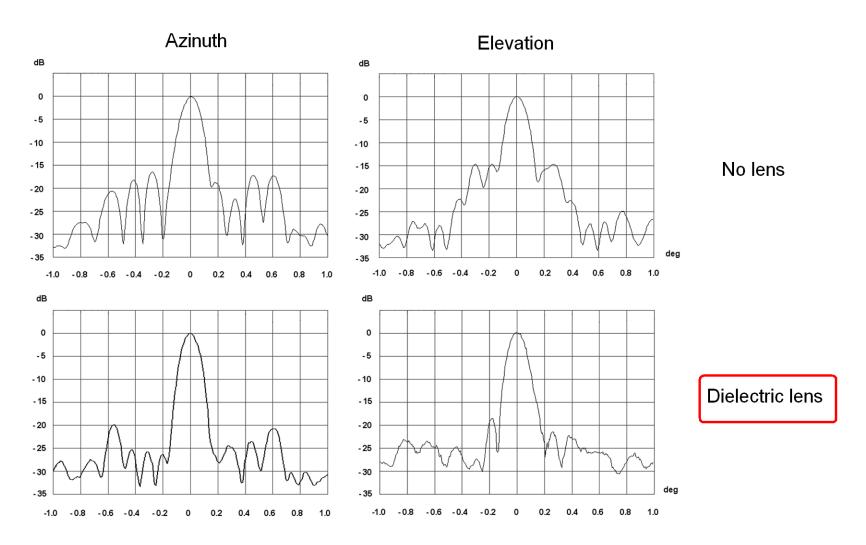
Ruze Equation

$$L_{R} = -685 \left(\frac{\text{rms}}{\lambda}\right)^{2} \text{ (dB)}$$

$$L_R = -685 \left(\frac{0.31}{3.9} \right)^2 = \left[-4.3 \text{ dB (gain loss)} \right]$$

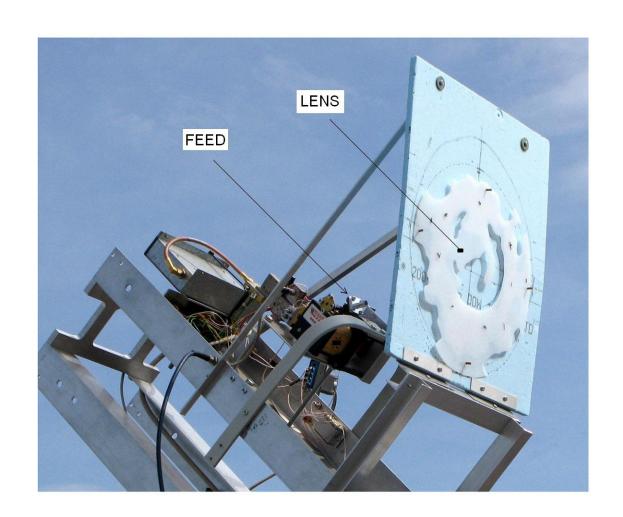
Feed in new position

Optimized after 3D scan



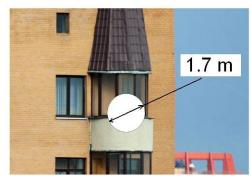
Feed and Lens

Beacon



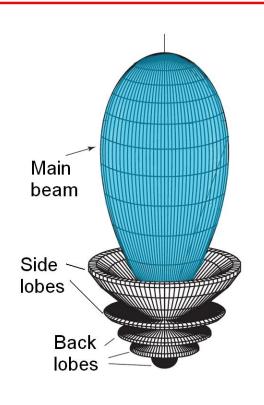


BW = 0.11° beam spot at 890 m distance

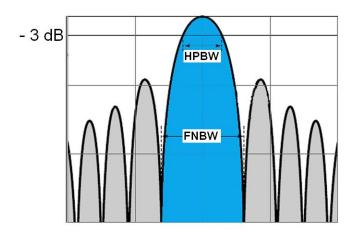


Main beam efficiency (ε_м)

The part of the total radiated (received) energy of an antenna that is contained in the main beam.



$$\varepsilon_{M} = \frac{\int_{MB} P_{n}(0,\Phi) d\Omega}{\int_{4\pi} P_{n}(0,\Phi) d\Omega}$$



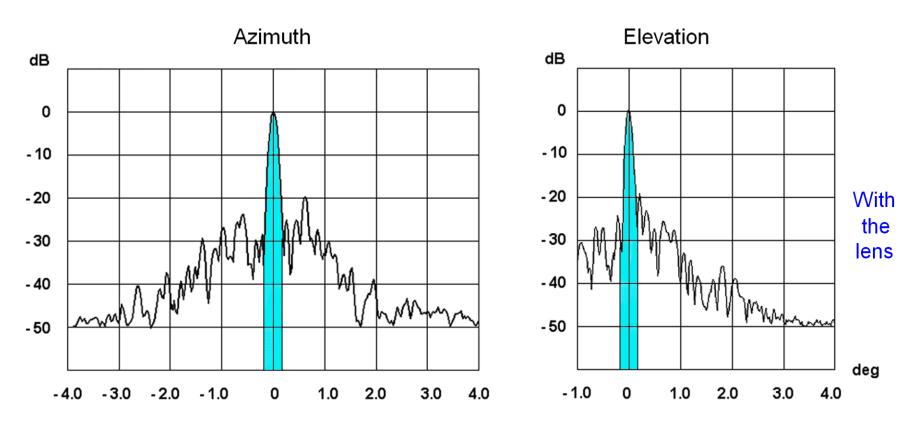
$$G_{_{A}} = \frac{4\pi A}{\lambda^{2}} \, \epsilon_{_{APP}} \, \eta \ = \pi^{2} \big(\frac{D}{\lambda}\big)^{2} \, k \, \epsilon_{_{M}} \, \eta \; ;$$

$$G_A \approx 8 \left(\frac{2400}{3.88}\right)^2 \epsilon_M ;$$

$$\varepsilon_{APP} = k * \varepsilon_{M}$$
; $k = 0.8$ for my DMH feed;

$$G_A \approx 3\,060\,000 * \epsilon_M$$

Antenna pattern for -50 dB noise floor



The integration of patterns gives the following results for $\varepsilon_{\rm M}$ – main beam efficiency:

Antenna without lens

 $\varepsilon_{\rm M} = 0.60$

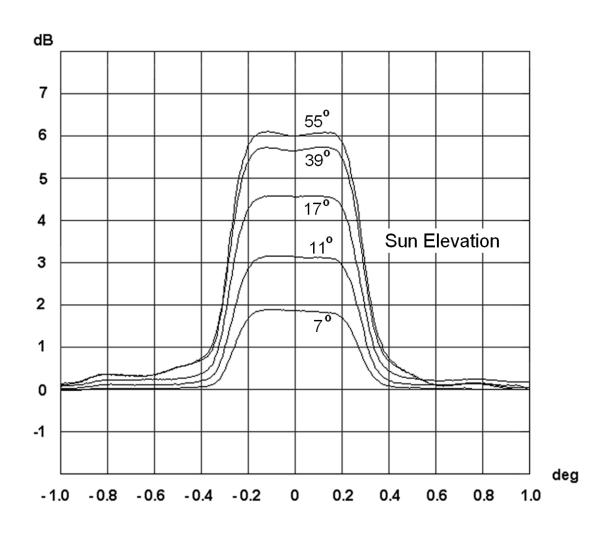
The lens benefit is approximately equal to 0.4 dB

Antenna with the lens $\epsilon_{\rm M}$ = 0.66

Beam efficiency and gain degradation -1.2 dB

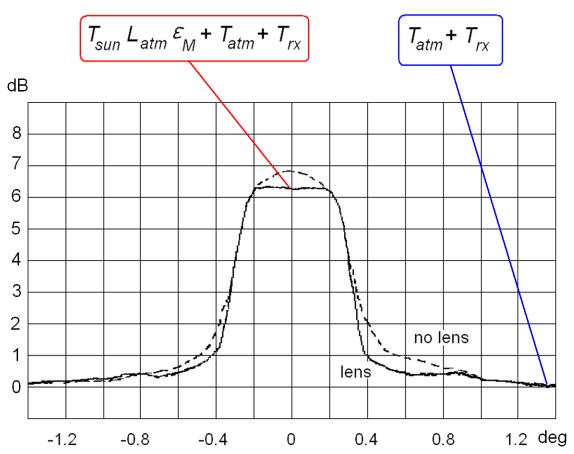
Calculated beam efficiency $\varepsilon_{\rm M} = 0.87$.

Sun noise (1)



with dielectric lens

Sun noise (2)



$$Y_{sun} = \frac{T_{sun} L_{atm} \varepsilon_M + T_{atm} + T_{rx}}{T_{atm} + T_{rx}}$$

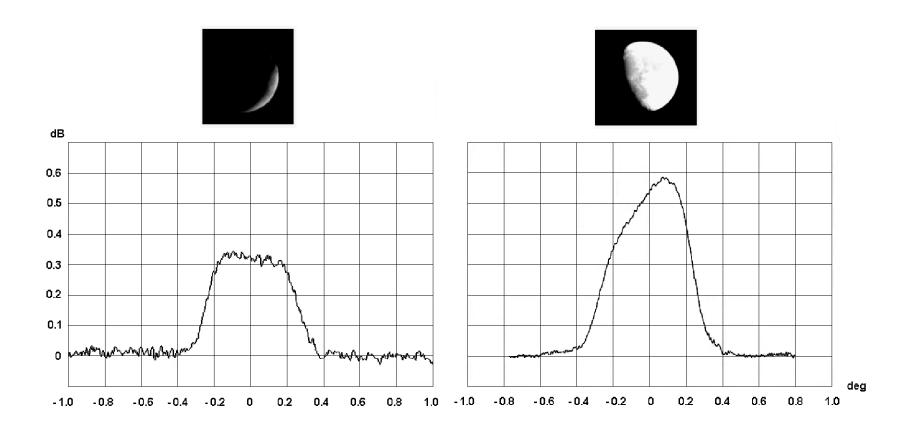
Main beam efficiency

$$\varepsilon_{M} = \frac{(T_{atm} + T_{rx})(Y_{sun} - 1)}{T_{sun} L_{atm}}$$

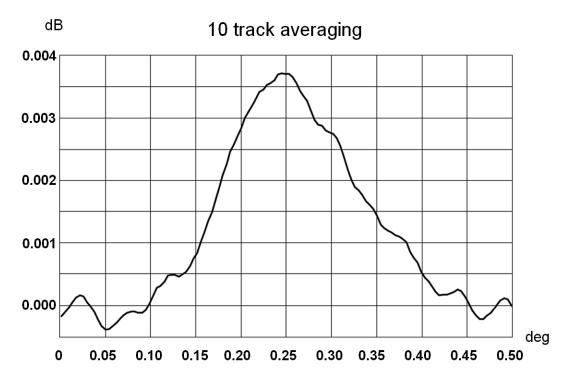
$$Y_{sun} = 4.3 (6.3 \text{ dB})$$
 $T_{rx} = 1030 \text{K}$
 $T_{atm} = 70 \text{K} \quad L_{atm} = 0.77$

$$\varepsilon_{M}^{} = 0.65 \quad T_{sun}^{} = 7250 \text{K}$$

Moon noise



Jupiter noise



Date 01.02.2014

Temperature -14°C

Humidity 50%

Ant elevation 45...50°

Sys noise temperature Radiometer RAD2 (LC Technologies) BW = 40 MHz

Antenna HPBW 400" Jupiter angular size 44"

Jupiter noise calculation:

$$T_{\text{JUPITER}} \approx T_{\text{NEW MOON}} \approx 140 \text{K}; \quad Y_{\text{NEW MOON}} = 0.33 \text{ dB};$$

$$Y_{\text{JUPITER}} = Y_{\text{NEW MOON}} * \left(\frac{44"}{400"}\right)^2;$$

CONCLUSION

- For 77GHz band EME you need expensive antenna with extremely high accuracy of the reflector surface and quite a large size.
- It is possible to use available low-cost antennas with necessary check of the antenna pattern.
- It is possible to compensate the surface deviations with help of special dielectric lens.
- The Sun is easy to use celestial source for antenna testing on 77GHz. Sun noise tests are very informative.
- Using Jupiter as a point source for EME antenna test is yet questionable.